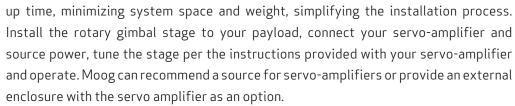


MODEL AC7332 GIMBAL STAGE

ROTARY SINGLE-AXIS GIMBAL



Moog has taken their expertise in high performance motion control components and elevated it in offering assemblies that combine many components into an integrated assembly. The single-axis gimbal stage combines our Matrix DC brushless direct drive motor and slip ring solutions with encoder, precision bearings and structure. Moog's gimbal stage is designed to support most payloads without the need for an external bearing system, thus reducing set-



The gimbal stage's physical configuration includes a bore. This allows for the installation of either a RF rotary joint, fiber optic rotary joint, optics pass, air channel or other device for operation in various systems. Also the bore allows Moog to upgrade a fielded gimbal stage to pass multiple high speed data channels as dictated by future system requirements without major redesign. For example, Moog can install a fiber optic rotary joint in the bore and provide data converters, media converters and / or multiplexers as a complete solution. This allows for easy and inexpensive upgrades, thus minimizing life cycle costs.

This gimbal stage is designed to be customized for unique applications without requiring major development expense. Moog has systems and components engineers in place to support your needs.









FEATURES

- All rotary components are integrated into one assembly
- Moog's direct drive brushless DC motor provides excellent torque, heat transfer and life characteristics
- Use of Moog's drum style slip ring technology assure proper transfer of electrical power and data signals
- Rugged precision encoder that is well proven in many environments
- Bearing system and structure are designed to support typical customer payloads and vibration profiles
- Designed for harsh environments
- Wide dynamic performance range
- Bore allows for integration of fiber optic rotary joint, RF rotary joint, optics, additional slip ring or air channel pass
- Excellent slew to queue and slow tracking performance
- Standard with flying leads. Options include cable with connectors or connector assembly
- Shaft and face seals are optional

APPLICATIONS

- Electro-optical / infrared sensor suites
- Radar and lidar pedestals
- Active countermeasures systems
- Specialized assembly, automation and industrial machinery
- Airborne, vehicular and stationary platforms
- Vision systems

BENEFITS

- Self-contained: easy installation
- · Supports many payloads without the need for an external bearing set
- Reliable and maintainable
- Excellent performance in small size and light weight
- Adaptable for use with high data rate sensors via installation of a fiber optic rotary joint or second slip ring without the need to replace the rotary gimbal stage assembly

		SPECIFIC	CATIONS		
Parameter	Units	Е	F	G	Notes
		Motor Pa	arameters		
Operational Temerature Range	°C		-54 to +71		
Relative Humidity		Up to 95% noncondensing			
Peak Stall Torque	N-m (lb-ft)	33.6 (24.8)	21.9 (16.2)	10.3 (7.60)	
Continuous Stall Torque	N-m (lb-ft)	11.4 (8.41)	5.3 (3.91)	1.3 (0.959)	@ 71°C and mounted to customer's heat sink
Travel		360° cont			
Maximum Axial Load (Z-axis)*	N (lb)	1,112 (250)	1,112 (250)	1,112 (250)	
Maximum Radial Load (X and Y axis)	N (lb)	1,112 (250)	1,112 (250)	1,112N (250)	
Maximum Moment (X and Y axis)	N-m (lb-ft)	54 (40)	54 (40)	54 (40)	
Stage Mass	Kg (lb)	8.0 (17.5)	6.25 (13.75)	5.25 (11.5)	Not including payload
Stage Rotational Inertia	Kg-m2 (ft-lb-s ²)	.009 (.007)	.008 (.006)	.007 (.005)	
OD (Body)	mm (in)	172 (6.75)	172 (6.75)	172 (6.75)	
OD (Flange)	mm (in)	203.2 (8.000)	203.2 (8.000)	203.2 (8.000)	Alternate mounting configurations available, contact factory
ID	mm (in)	35.6 (1.4)	35.6 (1.4)	35.6 (1.4)	
Axial Length	mm (in)	102 (4.00)	89 (3.5)	76.2 (3.00)	
Motor Type		Direct drive 3-p			
Rated Voltage	VDC	48	36	24	3 brushless to stage
Peak Current	ADC	20	20	20	3 brushless to stage
Continuous Current	ADC	7.5	6	5	3 brushless to stage
Number of Poles		32	32	32	
DC Resistance	Ohms	2.00	1.50	1.10	± 10%
Torque Sensitivity***	lb.ft / amp	1.31	0.875	0.439	± 10%
Back EMF Constant***	V per rad / s	1.78	1.19	0.595	± 10%
Inductance	mH	3.3	2.2	1.0	
		Encoder P	Parameters		
Туре		Rotary optical ab	Contact factory for other data formats		
Accuracy	Static pointing accuracy at 25°C of 0.01 degrees				Contact factory for higher accuracy
Resolution		26-bit (67,108,864	18 and 32 bit also available contact factory for other resolutions		
		Slip Ring F	Parameters		
Passes (Number of Rings)		29 signal rings and 2 power rings	17 signal rings and 2 power rings**	5 signal rings and 2 power rings	Total for power and data
Maximum Current	ADC	Power rings: 20 amps	Power rings: 20 amps**	Power rings: 20 amps	Maximum continuous
		Signal rings: 2 amps	Power rings: 2 amps**	Power rings: 2 amps	
Maximum Voltage	VDC	50	50**	50	Up to 50,000 feet altitude
Data	Twisted pair leads and ring diameter supports high speed digital data such as gigabit Ethernet**				

Notes:

Performance parameters are based on numeric modeling and are subject to change. Tolerances will be provided once the configuration is finalized.

* Maximum axial load, maximum radial load in one direction and maximum moment applied simultaneously would result in a calculated L10 bearing life of $2\,x\,10^7$ revolutions (-5,000 hours at 60 rpm)

** Example of F stage with 19 passes (rings): 2 power rings 20 A per ring 8 signal rings 1000BaseT Ethernet 4 signal rings RS-422 5 discrete / spare signal rings

***For frameless motor only: does not take into account frictical torques of the assembly

Other options available:

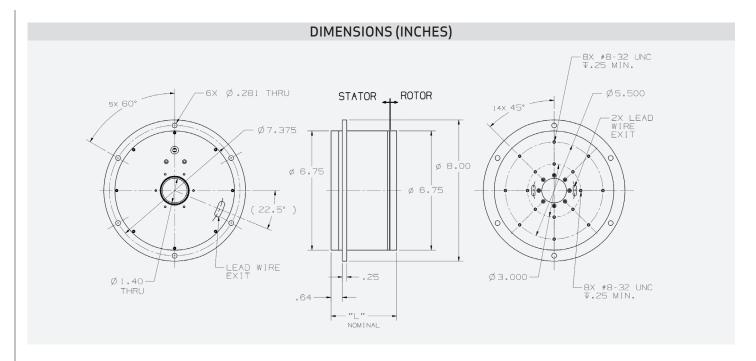
- Other options available:

 1. Alternative input voltages are available by providing a motor with a different winding configuration than shown, as follows:

 Model E: Input voltages up to 192 volts, 3 \(\phi\) brushless

 Model F: Input voltages up to 144 volts, 3 \(\phi\) brushless

 Model G: Input voltages up to 96 volts, 3 \(\phi\) brushless
- 2. Can be provided with seals
- 3. Slip ring variations are available within the number of passes shown, see ** above 4. Output plate configurations to accept various payloads



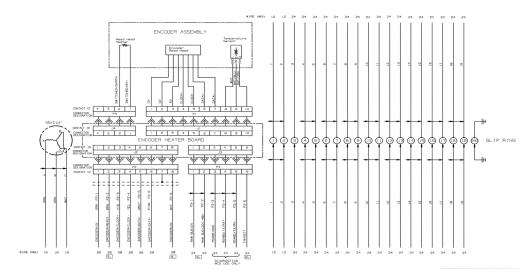
Stage Model	Dimension "A"		
Е	2.27 inches		
F	1.77 inches		
G	1.27 inches		

Notes:

- 1. When the rotor and stator slots are aligned as shown the encoder output is approximately in the zero position.
- 2. CW rotation of rotor as shown results in increasing encoder count.
- 3. Motor stepping sequence results in CW rotation as shown

STEPPING SEQUENCE						
Step	Green	Orange	White			
1	+		-			
2	+	-				
3		-	+			
4	-		+			
5	-	+				
6		+	-			

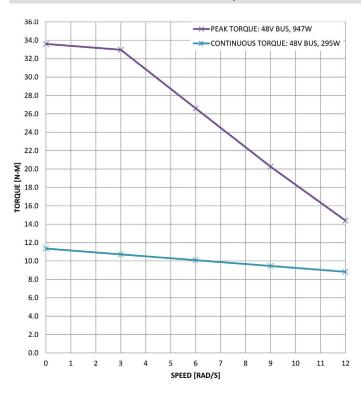
WIRING DIAGRAM (F STACK GIMBAL STAGE)



Notes:

- 1. Ground inhibit (P3-6) to encoder OV (WHT, P2-8) to disable encoder and heater. Leave inhibit floating to enable.
- 2. Encoder: ready (Brown lead P2-2) high indicates the encoder is powered on and ready. At temperatures below -40°C, the encoder will be heated for up to five minutes maximum before 5 V power is applied to the encoder. Heater power and encoder power are stepped down from 20 48 V raw bus power that is externally supplied.

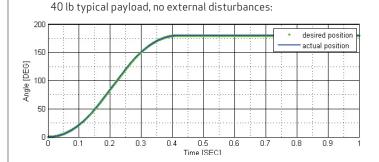
MODEL E, PERFORMANCE CURVES (@ 48 VDC AND @ 71 °C)

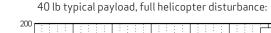


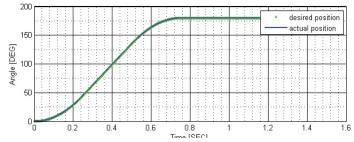
Notes

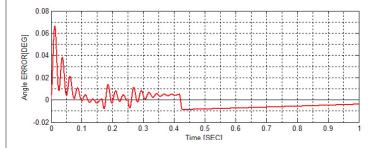
- 1. Maximum current of 20 A. Continuous current of 7.5 A
- Payload assumption. Gimbal stage oriented with axis of rotation vertical and payload directly above or below stage.
- 3. Payload is equivalent to a 20 lb aluminum flywheel that is 9 inches in diameter and 3.25 inches thick with a mass moment of inertia of .06 kg-m2. The 40 lb payload is equivalent to a 9 inch diameter 6.5 inch thick disk with a mass moment of inertia of .12 kg-m2.
- 4. Center of gravity (CG) is considered to be a maximum of 0.1 inches for the noted payload mass unless otherwise stated. Higher offset is allowable with a smaller payload mass.
- Performance curves are based on numerical modeling and subject to change. Contact Moog to determine exact performance in your system.

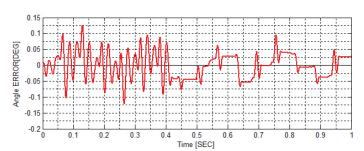
MODEL E, STEP INPUT PERFORMANCE CURVES





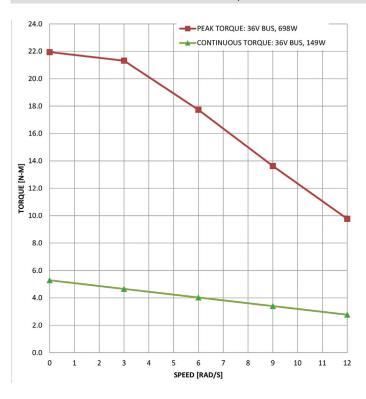






.02 g²/Hz vibe levels plus sine tones from a helicopter rotor

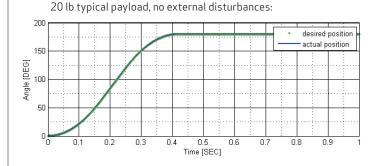
MODEL F, PERFORMANCE CURVES (@36 VDC AND @71°C)

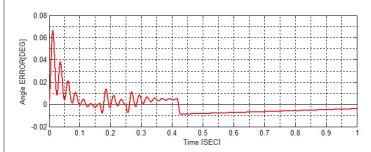


Notes

- 1. Maximum current of 20 A. Continuous current of 6 A.
- 2. Payload assumption. Gimbal stage oriented with axis of rotation vertical and payload directly above or below stage.
- 3. Payload is equivalent to a 20 lb aluminum flywheel that is 9 inches in diameter and 3.25 inches thick with a mass moment of inertia of .06 kg-m2.
- 4. Center of gravity (CG) is considered to be a maximum of 0.1 inches for the noted payload mass unless otherwise stated. Higher offset is allowable with a smaller payload mass.
- Performance curves are based on numerical modeling and subject to change. Contact Moog to determine exact performance in your system.

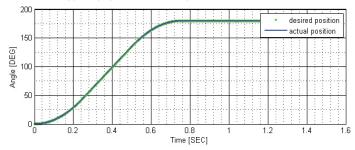
MODEL F, STEP INPUT PERFORMANCE CURVES

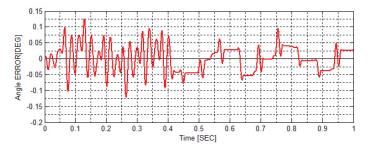




Maximun CG offset of 0.1 inches

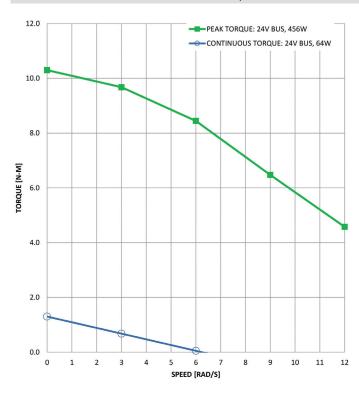
 $20\ lb\ typical\ payload, full\ helicopter\ disturbance:$





Maximum CG offset of 0.1 inches .02 g2/Hz vibe levels plus sine tones from a helicopter rotor

MODEL G, PERFORMANCE CURVES (@24 VDC AND @71°C)

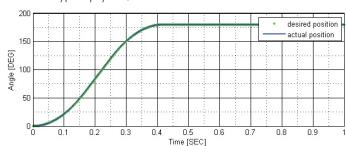


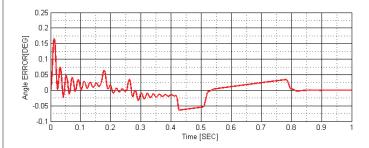
Notes.

- Maximum current of 20 A. Continuous current of 5 A. Payload assumption. Gimbal stage oriented with axis of rotation vertical and payload directly above or below stage.
- 2. Payload is equivalent to a 20 lb aluminum flywheel that is 9 inches in diameter and 3.25 inches thick with a mass moment of inertia of .06 kg-m2.
- 3. Center of gravity (CG) is considered to be a maximum of 0.1 inches for the noted payload mass unless otherwise stated. Higher offset is allowable with a smaller payload mass.
- Performance curves are based on numerical modeling and subject to change. Contact Moog to determine exact performance in your system.

MODEL G, STEP INPUT PERFORMANCE CURVES

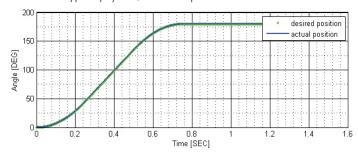
5 lb typical payload, no external disturbances:

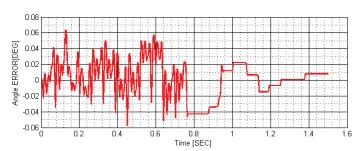




Maximun CG offset of 0.1 inches

20 lb typical payload, half helicopter disturbance:





Maximum CG offset of 0.1 inches .02 g2/Hz vibe levels plus sine tones from a helicopter rotor



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